

N91-17056

FLIGHT ELEMENTS

SPACE TRANSPORTATION AVIONICS

TECHNOLOGY SYMPOSIUM

FLIGHT ELEMENTS SUBPANEL

SUMMARY OF FINDINGS

CLAUDE R. KECKLER

NOVEMBER 9,1989

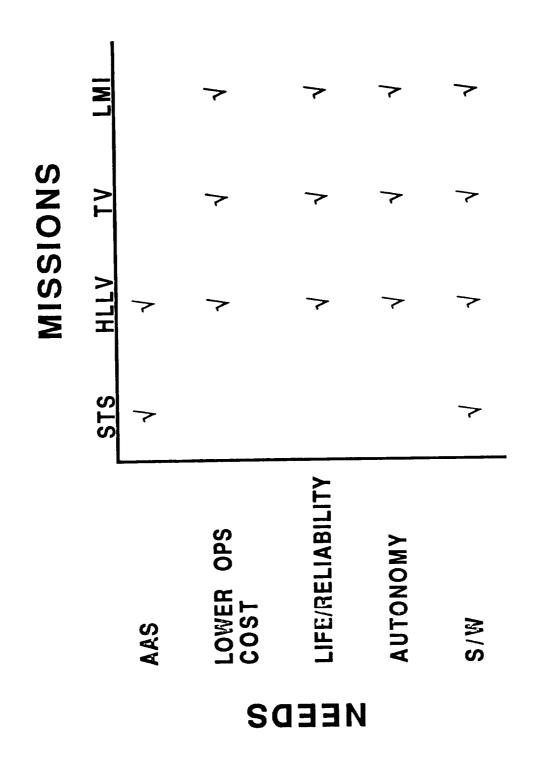
FLIGHT ELEMENTS

- ADVANCED AVIONICS ARCHITECTURE
- ADVANCED PROCESSORS
- · INTEGRATED GPS/GN&C
- · ADVANCED DISPLAYS & CONTROLS
- ADVANCED COMMUNICATION & TELEMETRY

- · LOVANCED SENSORS & INSTRUMENTATION
- · FAULT DETECTION & FAULT MANAGEMENT

• ADVANCED ELECTRICAL POWER DIST. & CONTROL

- · EMA/POWER SYSTEMS
- INFLIGHT CREW TRAINING

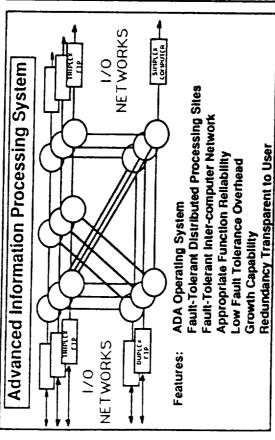


SPACE TRANSPORTATION AVIONICS TECHNOLOGY SYMPOSIUM

FLIGHT ELEMENTS

ADVANCED AVIONICS SYSTEMS ARCHITECTURES

NOVEMBER 1989



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- Improved Reliability at Lower Cost
- · Low Recurring Hardware and Operations Cost
- Enable/Support Launch-On-Demand
- · Open-Ended Architectures that Support System Growth and Change
 - · Vehicle-Wide Standardization of Architectural Concepts
- · Autonomous, Factory-To-Flight Subsystem Integrity and Confirmation
 - · Enable/Support Autonomous Long Duration/Distance Flight Operations
- Flexible/Secure Interfaces for Payload and Other Non-Avionic System Support
- Autonomous Pre-Filght System Support and Test

Major Milestones (1990-1995):

BAC - D. Johnson

JSC- Tom Barry, Tom Morre LaRC - C. Melssner, F. Pitts

Key Contacts:

CSDL - J. Lala

GD - J. Karas

MSFC - W. Chubb, W. Brantley

LeRC - H. Wimmer

WDRC - J. Stanley, R. Bortner

JPL - D. Rennels

TECHNOLOGY DEMO'S IN WORK:

- · MPRAS
- Common Module Military Aircraft Filght Tests

RECOMMENDED DEMOS:

- Define System Goals and P31 Planning (90 and 91)
- · Joint Lab Demo's at MSFC/JSC with FLT Test at Ames (92 and
- Insertion on Combined STS and Shuttle-C Upgrades

Facilities

- -JSC Avionics Eng. Lab
- -MSFC Avionics Productivity
- -LaRC AIRLAB Center

RIC - L. Shockley HI - J. Weyrauch

MMC - R. Gates

SPACE TRANSPORTATION AVIONICS TECHNOLOGY SYMPOSIUM FLIGHT ELEMENTS

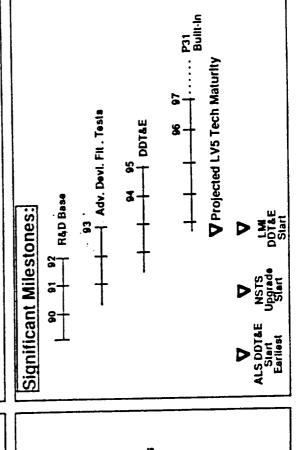
ADVANCED AVIONICS SYSTEMS ARCHITECTURES

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Technology Issues:

- · Level of Fault Tolerance
- · Cost vs. Reliability
- · Utility of Bullding-Block Architectures
- · Modeling/Test Mix for Validation
- · Design for Launch-With-Fallures
- EME-HARD Design and Assessment
- Software Development Environment
- · ADA Software for High-Bandwidth Control

Candidate Programs: - Assured Shuttle Availability, Unmanned Orbiter - NASP, CERV - Shuttle-C, ALS - Existing Launch Vehicles - SSP, Lunar Mars Intifative



Major Accomplishments:

- · Space Station Avionics Design Captures Some Objectives
- ALS Requirements and Advanced Technology Development Meets/Exceeds Objectives
- Advanced Military Aircraft in DDT&E (A-12 and ATF) Captures
 Objectives and Developing Usable Hardware
- Commercial aircraft fault-tolerant / distributed systems

SPACE TRANSPORTATION AVIONICS TECHNOLOGY SYMPOSIUM

NOVEMBER 1989 SEAMLESS HARDWARE AND SOFTWARE TRANSITIONS BETWEEN RELIABILITY FOR EXTENDED MISSIONS (1,000 - 10,000 HRS) · AUTONOMY TO ADAPT TO CHANGING SITUATIONS AND OFF-LINE COMPONENT LEVEL SELF TESTABILITY PROVIDE THE SYSTEM ARCHITECTURE TO ACHIEVE ON-LINE MODULE LEVEL VALIDATION **MAJOR MILESTONES (1990-1995):** · LOW POWER, WEIGHT, AND VOLUME HIGH PERFORMANCE (.1 TO 10 GOPS) FLIGHT SYSTEM PROTOTYPE 1995 **ARCHITECTURE DEFINITION 1990** BRASS BOARD PROTOTYPE 1993 LABORATORY PROTOTYPE 1991 ADVANCED AVIONICS ARCHITECTURE MAJOR OBJECTIVES: · RADIATION HARDNESS CONCEPT DEFINITION 1990 MISSION MODES **EPOCHS** FLIGHT ELEMENTS EXPLOIT THE POTENTIAL SYNERGISM BETWEEN PARALLEL PROCESSING AND REDUNDANCY **ADVANCED AVIONICS CONCEPTS** · T. DE YOUNG (DARPA) KEY CONTACTS: · B. J. THOMAS (IBM) · J. DEYST (CSDL) · H. BENZ (LARC)

SPACE TRANSPORTATION AVIONICS TECHNOLOGY SYMPOSIUM

FLIGHT ELEMENTS

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ADVANCED AVIONICS ARCHITECTURE | CANDIDATE PROGRAMS:

TECHNOLOGY ISSUES:

INTERCONNECTION TOPOLOGY

 PARALLEL COMPUTATION THROUGHPUT OVERHEADS

• INFORMATION TRANSFER • FAULT TOLERANCE

FUTURE AUTONOMOUS SPACECRAFT

LUNARMARS INITIATIVE

NASP

SOFTWARE

• OPERATING SYSTEM
• REDUNDANCY MANAGEMENT

QUANTIFIABLE PERFORMANCE AND RELIABILITY

VALIDATION METHODOLOGY

LOW POWER/SMALL FEATURE SIZE/RADIATION HARDNESS

SIGNIFICANT MILESTONES:

91 | 92 | 93 | 94 | R&D BASE

93 | 94 | 95 | 96 | ADV. DEVEL.

95 | 96 | 97 | 98 | FLIGHT SYSTEM

MAJOR ACCOMPLISHMENTS:

RECOGNITION OF THE NEED FOR SUCH SYSTEMS.

SPACE TRANSPORTATION AVIONICS TECHNOLOGY SYMPOSIUM FAULT DETECTION AND FAULT MANAGEMENT FLIGHT ELEMENTS

TECHNOLOGY CONCEPT:

Sensors

Reconfiguration

Design Knowledge Capture

Operations

Software

AN INTEGRATED APPROACH FAULT MANAGEMENT:

FDIR

Diagnosis

Detection/Isolation

Hardware

Testability

Digraph Matrix Analysis

Performance

KEY CONTACTS:

Industry contacts: TBD ARC - A. Patterson-Hine

LaRC - C. Meissner JSC - J.T. Edge

MSFC - D. Weeks

KSC - T. Davis JPL - D. Miller

HQ - G. Swletek (OSS), J. Di Battista (OAST)

KEY FACILITIES:

JSC Testbeds

ARC Advanced Architectures Testbed MSFC SSM/PMAD & ECLSS Testbeds

MAJOR OBJECTIVES:

- Monitoring, diagnosis, and reconfiguration at all system levels
- Unambiguous Isolation of failures
- · Integration with maintenance support and operations
 - Optimize system operations to manage degraded system performance;
 - Lower development/operations costs
- Develop fault tolerant/FDIR requirements and specifications

MAJOR MILESTONES:

- · Review technology, investigate leveraging opportunities (1990)
- program technology development and Define concept and develop integrated integration plan(1990)
- Develop integrated testbed(s) (1992)
- Proof of concept demo (1993)

SPACE TRANSPORTATION AVIONICS TECHNOLOGY SYMPOSIUM FAULT DETECTION AND FAULT MANAGEMENT FLIGHT ELEMENTS

TECHNOLOGY ISSUES:

CANDIDATE PROGRAMS:

· SSFP

· ALS

- Design accomodation of fault detection and fault management (FD/FM)
 - Integrated program database support of FD/FM
 - Evolutionary, automated modeling techniques Design knowledge capture to support FD/FM
 - Scalability of current technologies
- Scope of human interface/interaction Software FD/FM

Lunar/Mars missions

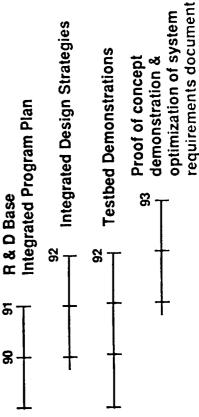
Shuttle C

- processing functions with high reliability and lower Development of smart sensors and specialized power consumption
 - Autonomous detection and recovery from faults \sqrt{q} i d $\sigma \neq i s \cap c$

MAJOR ACCOMPLISHMENTS:

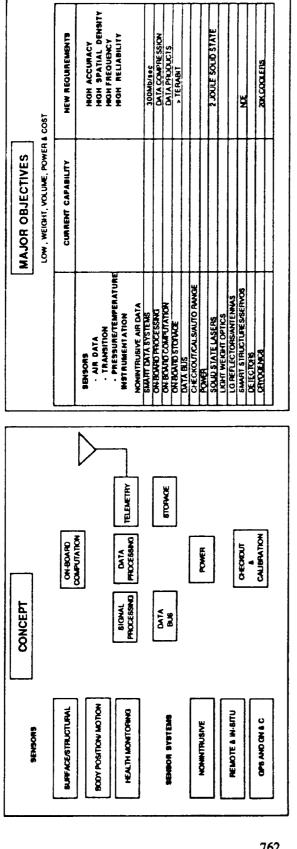
- already addressing some of the technology issues Space Station Advanced Development Program
- DARPA and ONR activities leveraged to some of the technology issues
- Basic testbeds already in place
- · Core Technology Team available

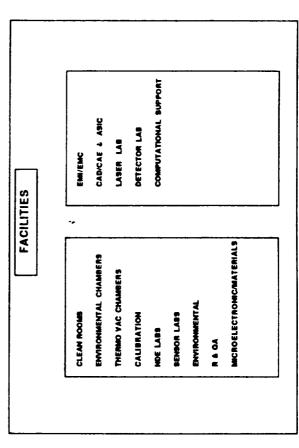
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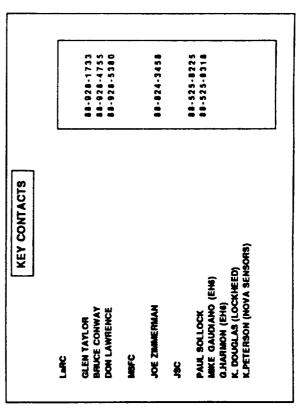


SPACE TRANSPORTATION AVIONICS TECHNOLOGY SYMPOSIUM FLIGHT ELEMENTS

ADVANCE SENSORS & INSTRUMENTATION

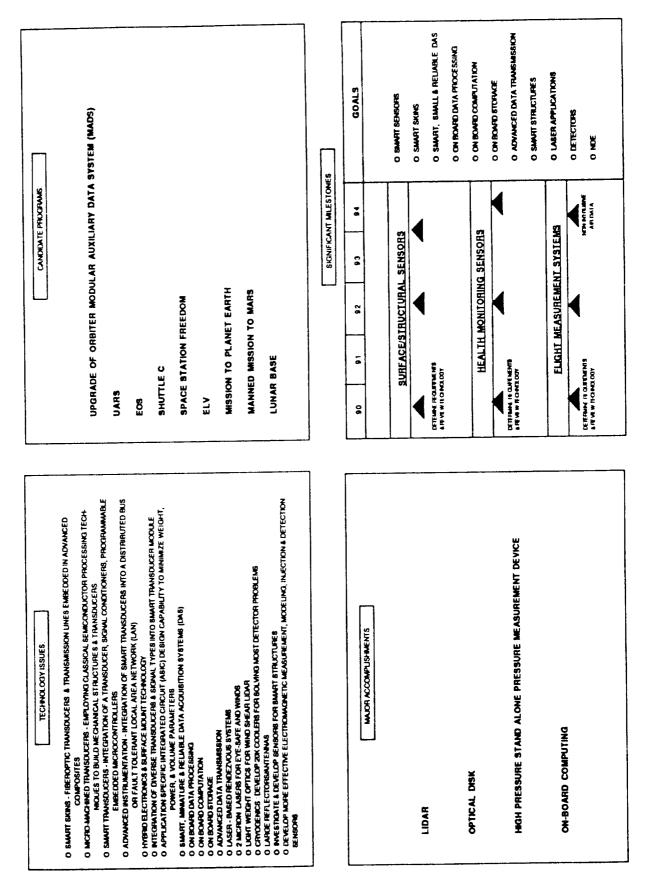




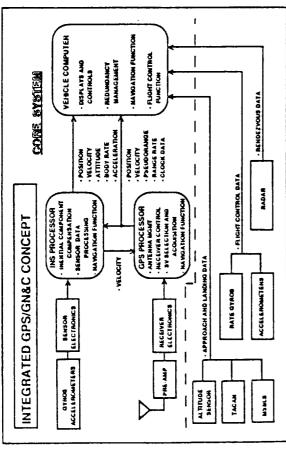


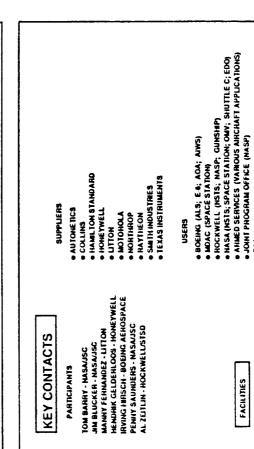
SPACE TRANSPORTATION AVIONICS TECHNOLOGY SYMPOSIUM FLIGHT ELEMENTS

ADVANCE SENSORS & INSTRUMENTATION



INTEGRATED GPS/GN&C





REDUCE MANTEMANCE COSTS REDUCE MANTEMANCE COSTS REDUCE URU COUNT HIGHEN MIBS AVOID OBSOLESCENCE OF DELETED SYSTEMS REDUCE VEHICLE LAUNCH AND TURNAROUND TIME TESTING - CALIBRATION - ALIGNMENT DEVELOP COMMON MODULAN SYSTEMS FOR MULTIPLE NASA APPLICATIONS PROVIDE AUTOLAND CAPABILITY ASCENT - ORBIT - HENTRY REDUCE GROUND SUPPORT REDUCE GROUND SUPPORT PROVIDE ATTITUDE DETERMINATION CAPABILITY REDUCE GROUND SUPPORT REDUCE GROUND SUPPORT

MAJOR MILESTONES

- STANDARD RLG INS AND GPS INTEGRATED SYSTEMS DELIVERED TO NAVY AND AIR FORCE
- INTEGRATED GPSANS SYSTEMS DELIVERED FOR AF RC-135 AIRCRAFT
- INTEGRATED GPSAMS SYSTEM FOR REMOTELY PLOTED VEHICLE SUCCESSFULLY TESTED
- HELICOPTER AND ARCRAFT LANDING TESTS USING DIFFERENTIAL GPS SYSTEMS

CONDUCTED BY NASA-AMES

INS WITH EMBEDDED GPS RECEIVER IN PRODUCTION FOR CIVIL AVIATION (FOREIGN)

- HIGH PRECISION ORBIT NAVIGATION FILTER (KALMAN) DEVELOPED BY NASA JSC
- RELATIVE NAVIGATION CAPABILITY FOR RENDEZYOUS OPERATIONS INVOLVING
 TWO VEHICLES WITH GPS RECEIVERS EVALUATED BY NASAJSC

• JET PHOPOLUSION LABORATORY

NASA JSC GPS LABORATORY
 AF GEOPHYSICS LABORATORY

DARPA (GPS GUIDANCE PACKAGE)

GPS JPO GPS STANDAND AND PRECISE POSITIONING SERVICE
 NASA AMES MOBILE DIFFERENDAL GPS GROUND FACILITY
 OTHER GUVERNMENT AND CONTRACTOR FACILITIES

INTEGRATED GPS/GN&C

TECHNOLOGY/APPLICATION ISSUES

- ACQUISITION OF TARGET VEHICLE DATA FOR AUTONOMOUS NAVIGATION DURING RENDEZVOUS, PROXIMITY AND DOCKING OPERATIONS
 COOPERATIVE STREET
 'ANCHORY SATELLIFE
- VEHECLE ATTITUDE DETERMINATION USING GPS
 ANTENNA SEPARATION LIMITED BY VEHICLE DIMENSIONS
- TRACKING SATELLITE VEHICLE SIGNAL THROUGH PLASMA
- MEETING AUTOLAND PERFORMANCE REQUIREMENTS
 ACCURACY OF ALTITUDE DATA
- GPS UTLIZATION ABOVE 11,000 NM (e.g.; LUNAR MESSION RETURN)
 REDUCED SATELLITE VEHICLE VISHBILITY

MAJOR ACCOMPLISHMENTS

- F-15 FLIGHT TESTS DEMONSTRATE INFRITAL NAVIGATION ASSEMBLY CAPABILITY
 TO PROVIDE NAVIGATION AND FLIGHT CONTROL DATA 1948
- INTEGRATED INS WITH EMBEDDED GPS FLOWN IN BOEING 767
 FLIGHT TESTS PROGRAM 1888
- FAA CERTIFICATION OF INS WITH EMBEDDED GPS LATE 1990
- NATIONAL AERO SPACE PLANE SUBSYSTEM CONSORTIUM INVESTIGATING ANTENNA DESIGNS, ADVANCED ELECTROMICS, PLASMA TRANSMITRECEIVE LIMITATIONS
- SHUTTLE INTEGRATED GPS-GNAC CONCEPT AND FEASIBILITY STUDY STUDY COMPLETE: 1860 FLIGHT DEMONSTRATION - 1860
- PRELIMINARY DESIGN STUDY OF INTEGRATED GPS & INS (MSFC)
 INTEGRATED AND SEPARATE INSIGPS
 MODULARIZED CONFIGURATION
- LABORATORY SIMULATIONS AND EVALUATIONS ON-ORBIT OPERATIONS AUTOHOMOUS NAVIGATION AUTOLAND

CANDIDATE PROGRAMS

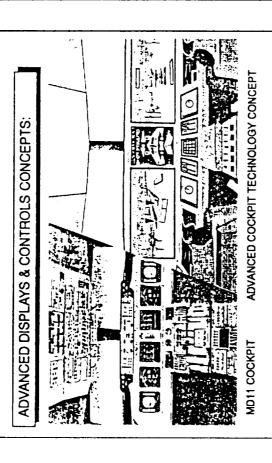
- ASSURED SHUTTLE AVAILABILITY (ASA)
- SHUTTLE C
- EXTENDED DURATION ORBITER (EDO)
- ASSURED CREW RETURN VEHICLE (ACRV)
- SPACE STATION
- ORBITAL MANEUVERING VEHICLE
- ADVANCED LAUNCH STAGE
- ADVANCED UPPER STAGES
- NATIONAL AERO SPACE PLANE (NASP)
- LUNAR AND MARS MISSIONS RETURN

SIGNIFICANT MILESTONES

- IMPLEMENT STANDARD, MODULAR GPS RECEIVER
 COST EFFECTIVE
 SUPPORTS MULTIPLE PROGRAMS
 CONFIGURABLE TO SPECIFIC APPLICATION
 MICLUDE TESTABLITY AS DESIGN REQUIREMENT
- CONDUCT TRADE STUDY OF TECHNIQUES TO ACCOMPLISH AUTOLAND, INCLUDING A FLIGHT DEMONSTRATION
- CONDUCT TRADE STUDY OF TECHMONES TO PERFORM AUTONOMOUS NAVIGATION, BY MISSION PHASE, FOR VARIOUS TRANSPORTATION PROGRAMS ASCENT - ORBIT - REENTRY
- CONDUCT TRADE STUDY OF TECHNIQUES FOR GPS DETERMINATION OF VEHICLE ATTITUDE, INCLUDING A FLIGHT DEMONSTRATION

SPACE TRANSPORTATION AVIONICS TECHNOLOGY SYMPOSIUM FLIGHT ELEMENTS

ADVANCED DISPLAYS AND CONTROLS



KEY CONTACTS:

- DEAN KOCIAN/WRIGHT R&D CENTER/SUPERCOCKPIT PROGRAM
 - DOC DOUGHERTY/ DARPA/ PILOT'S ASSOCIATE PROGRAM
- FRANK GOMER/HONEYWELL/PHEONIX RESEARCH CENTER
- GENE ADAW/ McDONNEL DOUGLAS/ 'BIG PICTURE' DISPLAY PROGRAM ANDREW FARKAS/ JOHNSON SPACE CENTER/ EF2
- DR. Mogreevy/ Ames research centery Aerospace Human Factors DIV.
- BILL RUCKS/ ROCKWELL STSD
- TERRY EMERSON/WRIGHT R&D CENTER/COCKPIT INTEGRATION DIRECTORATE

FACILITIES:

- JSC/ EF2 D & C PORTION OF ADV. SYSTEMS DEVELOPMENT LAB
- JSC/SHUTTLE ENGINEERING SIMULATOR
- LAPC/ADV. CONCEPTS SIMULATOR & CREW STATION SYSTEMS RESEARCH LAB
- LARC/TRANSPORT SYSTEMS RESEARCH VEHICLE (AFT FLT. DECK W/COLOR DISPLAYS)
 - APC/MAN VEHICLE SYSTEMS RESEARCH FACILITY & FLIGHT SIMULATION COMPLEX
 - WRIGHT R & D CENTER/ SUPERCOCKPIT LAB & "WAGIC" COCKPIT FACILITY

MAJOR OBJECTIVES:

- LOWER COST, IMPROVED MAINTAINABILITY/RELIABILITY
- FOR SHUTTLE IN PARTICULAR, ELIMINATE PARTS/SKILLS OBSOLESCENCE
- REDUCED WEIGHT, VOLUME, AND POWER CONSUMPTION
- INHERENT GROWTH CAPABILITY FOR NEW FUNCTIONS OR ADVANCING TECHNOLOGY (I.E., PAYLOAD USER I/F)
 - IMPROVEMENT IN PILOT'S SITUATIONAL AWARENESS
 - REDUCTION IN PILOT'S/OPERATOR'S WORKLOAD
- IMPROVED FLIGHT SAFETY AND OPERATING EFFICIENCY COMMONALITY AND SOFTWARE RECONFIG. INTERFACE FOR FLEXIBILITY AND LOWER COST IN THE SUPPORT OF MULTIPLE PROGRAMS
 - ELIMINATION OF PAPERMANUAL CLUTTER THROUGH USE OF INTERACTIVE OPTICAL DISK TECHNOLOGY
- IMPROVED AUTONOMY THROUGH USE OF AI AND HUMAN-CENTERED AUTOMATION

MAJOR MILESTONES (1990 - 1995)

SIGNIFICANT IMPROVEMENTS IN FLAT-PANEL TECHNOLOGIES

FY 90-91	FY 91-92
— FULL-COLOR, SUNLIGHT-LEGIBLE LIQUID CRYSTAL DISPLAYS	— FULL-COLOR PLASMA PANEL (15-IN. DIAG.), PHASE II SBIR

- FY 91-92 FY 91-92 FLIGHTWORTHY GRAPHICS GENERATORS CAPABLE OF REAL-WORLD 3-D PICTORIAL DISPLAYS
 - IMPROVEMENTS IN VOICE, TOUCH, AND HAND-CONTROLLER INPUT TECHNOLOGIES
- FY 91-92 FY 92-93 RESULTS OF AIR FORCE SUPERCOCKPIT AND BIG PICTURE PROGRAMS FINALIZED SPACE STATION MULTI-PURPOSE APPLICATIONS CONSOLE DESIGN
- FY 93-94 FY 93-94 RESULTS OF DARPA PILOT'S ASSOCIATE AND HDTV STUDIES
 - RESULTS OF NASA AIRCRAFT SAFETY/AUTOMATION PROGRAM ORBITER GLASS COCKPIT DISPLAY UPGRADE

FY 93-95

88:

SPACE TRANSPORTATION AVIONICS TECHNOLOGY SYMPOSIUM FLIGHT ELEMENTS ADVANCED DISPLAYS AND CONTROLS

TECHNOLOGY ISSUES:

- ORBITER DOWN-TIME FOR HARDWARE INSTALLATION
- MATURITY OF FLAT-PANEL DISPLAY TECHNOLOGY
- DANGER OF MAKING CREW BORED/MACHINE MINDERS
- ADVANCED DISPLAY SYMBOLOGY/PICTORIAL FORMATS
- MATURITY AND UTILIZATION OF AI TECHNOLOGY
- GROWTH AND FLEXIBILITY
- INTERACTIVE DISPLAY/CONTROL NEEDS MORE RESEARCH
- IMPACT OF ELECTRONIC DISPLAYS & CONTROLS (ALL-GLASS COCKPIT) ON CREW TRAINING

MAJOR ACCOMPLISHMENTS:

- EMERGENCE OF SEVERAL GLASS COCKPITS IN MILITARY AND COMMERCIAL AIRCRAFT (747-400, GULFSTREAM G IV, MD11, F·15E, AND BEECH STARSHIP)
- EMERGENCE OF COLOR ACTIVE-MATRIX LCD TECHNOLOGY •
- EMERGENCE OF HIGH-DEFINITION TV (HDTV) TECHNOLOGY

EMERGENCE OF REAL-TIME GRAPHICS DISPLAY TECHNOLOGY

EMERGENCE CONTINUOUS-SPEACH, SPEAKER-INDEPENDENT VOICE RECOGNITION TECHNOLOGY

CANDIDATE PROGRAMS:

- SPACE SHUTTLE (ASSURED SHUTTLE AVAILABILITY)
- SPACE STATION MPAC
- NATIONAL AERO-SPACE PLANE
- COMBINED AFT MANIPULATOR WORKSTATION (ORBITER)
- AVIATION SAFETY/AUTOMATION
- ADVANCED COCKPIT/FLIGHT MANAGEMENT TECHNOLOGY (PROPOSED FY 92 NEW INITIATIVE IN AERO)

SIGNIFICANT MILESTONES:

- 1995-96 SPACE STATION PERMANENT MANNED CAPABILITY, MPAC
- CREW EMERGENCY RETURN VEHICLE

1997

- MANNED LUNAR MISSION
- MANNED MARS MISSION

2016 2001 ORBITER BLOCK II COCKPIT

TRANSPORTATION AUTONICS TECHNOLOGY SYMPOSIUM FLIGHT ELEMENTS SPACE

ADVANCED COMMUNICATIONS AND TELEMETRY

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ADVANCED TECHNOLOGY

- GALLIUM ARSENIDE VHSIC
 - 2. FIBER OPTICS
- 3. ADVANCED ANTENNA DESIGN
- 4. FREE SPACE OPTICAL COMMUNICATION
- 5. ADVANCED SIGNAL PROCESSING 6. ADVANCED MODEM / CODEC DEVELOPMENT

MAJOR OBJECTIVES:

- UTILIZE NEW SPECTRUM
- MAXIMIZE DATA RATE THROUGH AVAILABLE SPECTRUM PROVIDE FLEXIBLE WIDEBAND DATA DISTRIBUTION
 - NETWORKS (DDNs)
 - VERY LOW POWER CONSUMPTION
 - DENSE PACKAGING
 - **RFVEMINMUNITY**
- GRACEFUL DEGRADATION
 - **MULTIBEAM ANTENNAS**

MAJOR MILESTONES:

DARPA MIMIC PHASE I (MAY'89), PHASE II (1991-94)

EVOLUTION OF STANDARDS

- FDDI STANDARD

32 GHz TWTA 7W 1992, 50W 1995 **ACTS COM SYSTEM**

"COMMON" SIGNAL PROCESSOR (CSP, GSP, EMSP, GASP)

GSFC/ M. FITZMAURICE, D. DALTON NMSU/ F. CARDEN, S. HORAN

LeRC/ R. LEONARD, J. HARROLD

JSC/ K. LAND

KEY CONTACTS:

LORC: MMW TEST FACILITY; DSP LAB. JSC- C&T ENGINEERING LAB.

NMSU- CENTER FOR SPACE TELEM. &

GSFC: LASER COM. LAB.

TELECOM. SYSTEMS

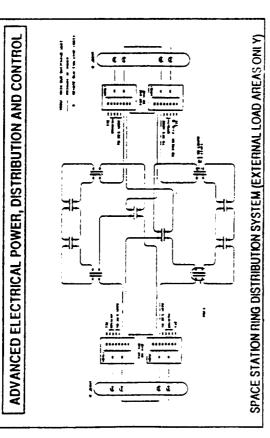
SPACE TRANSPORTATION AUIONICS TECHNOLOGY SYMPOSIUM FLIGHT ELEMENTS ADDANCED COMMUNICATIONS AND TELEMETRY

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CANDIDATE PROGRAMS: - SPACE STATION - STS UPGRADES - LUNAR MARS EXPLORATIONS - ATDRSS	SIGNIFICANT MILESTONES: - SMALLER, LIGHTER, LOWER POWER PACKAGING - IMPROVED RELIABILITY - STANDARDIZATION OF INTERFACES
1. PRODUCIBILITY OF G&AS 2. POWER LIMITS ON DISTRIBUTION 3. PACKAGING @ HIGH (> 20 GHz) 4. POINTING ACCURACY' STABILITY 5. SOFTWARE DEVELOPMENT 6. SOFTWARE DEVELOPMENT	MAJOR ACCOMPLISHMENTS: - 32GHz PHASED ARRAY UNDER DEVELOPMENT - Gbpæ FIBER OPTICS LINKS IN LABORATORY TEST - VHSIC PHASE I CHIPS AVAILABLE

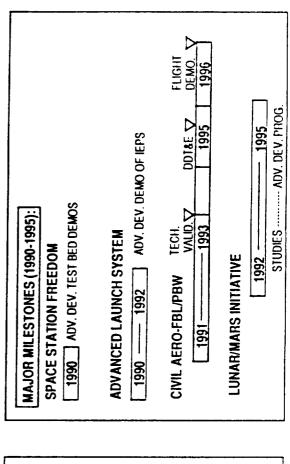
SPACE TRANSPORTATION AVIONICS TECHNOLOGY SYMPOSIUM ADVANCED ELECTRICAL POWER, DISTRIBUTION AND CONTROL FLIGHT ELEMENTS

NOVEMBER 1989



MAJOR OBJECTIVES:

REDUCE COSTS TO LEO, LUNARMARS SURFACE
REDUCE WEIGHT
INCREASE AVAILABLE POWER/ENERGY
IMPROVED REDUNDANCY MANAGEMENT
IMPROVED POWER QUALITY, USER AVAILABILITY
FAULT TOLERANT, INTEGRATED BITE



H. BRANDHORST/Lerc

KEY CONTACTS:

SPACE TRANSPORTATION AVIONICS TECHNOLOGY SYMPOSIUM **ADVANCED ELECTRICAL POWER, DISTRIBUTION AND CONTROL** FLIGHT ELEMENTS

NOVEMBER 1989

TECHNOLOGY ISSUES:

CANDIDATE PROGRAMS:

END-TO-END EPS MANAGEMENT WITH FAULT LIMITING, RECOVERY AND FAIL SAFE/FAIL OPERATIONAL RECONFIGURATION

DISTRIBUTED vs. DEDICATED PMAD FOR REDUNDANCY, RELIABILITY, OPERABILITY

BITE INTEGRATED INTO DESIGN AT MANUFACTURE

ASA: DDT&E FOR ELECTRICAL ACTUATORS RETROFIT BY INSPECTION DATE

MAJOR ACCOMPLISHMENTS:

- DEMONSTRATED MULTI-REDUNDANT, FAULT TOLERANT, MICROPROCESSOR CONTROLLED SSF 20 KHZ ELECTRICAL POWER DISTRIBUTION SYSTEM
- DEMONSTRATED VARIABLE SPEED DRIVES TO 200 HP, ELECTRICAL ACTUATORS TO 25 HP/DESIGNS TO 75 HP

ADVANCED LAUNCH SYSTEM ASSURED SHUTTLE AVAILABILITY CIVIL AERO - POWER-BY-WIRE/FLY-BY-LIGHT LUNAR/MARS INITIATIVE AF/WRDC - MORE ELECTRIC AIRPLANE - RETROFIT F-16

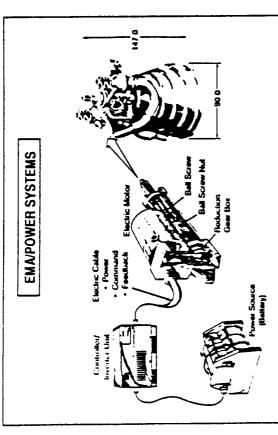
DAVID TAYLOR SHIP R&DC - ELECTRONIC NAVY

SIGNIFICANT MILESTONES: [1990] R&T BASE - COMPS, POWER SEMI'S [1991] 1992] ADV. DEV. - SSF, ALS [1991] 1992] ADV. DEV. - SSF, ALS [1991] 1992] ADV. DEV. - SSF, ALS [1992] ADV. DEV. - SSF, ALS [1993] ADV. DEV. - SSF, ALS [1994] ADV. DEV. - SSF, ALS [1994] ADV. DEV. - SSF, ALS [1995] ADV. DEV. - SSF, ALS [1996] ADV. DEV. - SSF, ALS [1996] ADV. DEV. - SSF, ALS [1996] ADV. DEV. - SSF, ALS [1997] ADV. DEV. - SSF, ALS [1998] ADV. DEV. - SS

 ADVANCED MOTOR CONTROL ENABLING INDUCTION MOTOR EXPLOITATION FOR LUNAR/MARS VEHICLES

SPACE TRANSPORTATION AVIONICS TECHNOLOGY SYMPOSIUM **EMA/POWER SYSTEMS FLIGHT ELEMENTS**

NOVEMBEF 1989



REDUCE KSC TURN AROUND COSTS; INCREASE LAUNCH RATI: · MATCH FLIGHT CONTROLS, POWER SOURCE, ACTUATORS MPROVE REDUNDANCY, RELIABILITY AND DECREASE WEIGHT USE DEMAND DRIVEN SYSTEM - SIMPLE IMPLEMENTATION - ELIMINATE EXCESSIVE MAN TESTS AND VERIFICATIONS - ELIMINATE GROUND SUPPORT CARTS AND EQUIPMENT - ADD SELF CHECKOUT THROUGH BUILT-IN-TEST (BITE) FECHNOLOGY TRANSFER TO CIVIL SECTOR AUTOMATED VEHICLE CHECKOUT LOW STANDBY POWERVENERGY - ELIMINATE HYDRAULIC SILTING MPROVE DISPATCH RELIABILITY REDUCE STANDDOWN TIME MAJOR OBJECTIVES:

DEMO. O DOTAE 1990 TECHNOLOGY, RISK, COST ASSESSMENT 1994 COMPLETE DEMOS/ \$\square\$ COSTAPERABILITY\$ ASSURED SHUTTLE AVAILABILITY (ASA) 1993 **MAJOR MILESTONES (1990-1995):** ADVANCED LAUNCH SYSTEM VALID, STUDIES **THEVIEW TECH** 1992 CIVIL AERO-FBL/PBW **TEMA DESIGNS** 1991 1991 1990

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TECHNOLOGY ISSUES:

ASA: DDT&E FOR EMA RETROFIT BY INSPECT. DATE

END-TO-END EPS MANAGEMENT WITH FAULT LIMITING, RECOVERY AND FAIL SAFE/FAIL OPERATIONAL RECONFIGURATION

DISTRIBUTED vs. DEDICATED PMAD FOR REDUNDANCY, RELIABILITY, OPERABILITY

BITE INTEGRATED INTO DESIGN AT MANUFACTURE

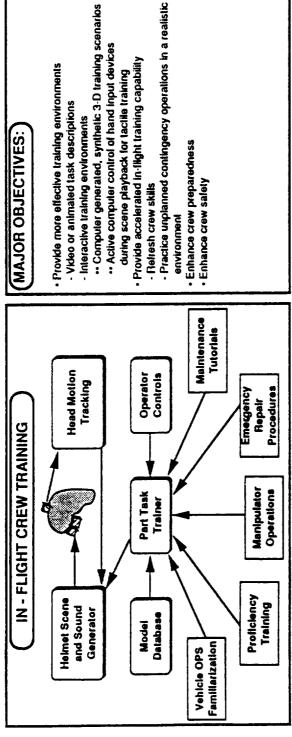
MAJOR ACCOMPLISHMENTS:

- PRELIMINARY ASA STUDIES COMPLETED
- KSC TURNAROUND FLOW ANALYSIS INITIATED
- TECHNOLOGY DEMOS/ASSESSMENT INITIATED
- DEMONSTRATED ELECTRIC ACTUATORS/ DRIVES TO 25 HP/DESIGNS TO 75 HP
- DEMONSTRATED MULTI-REDUNDANT, FAULT TOLERANT, MICROPROCESSOR CONTROLLED SSF 20 kHz ELECTRICAL POWER DISTRIBUTION SYSTEM

				٥	LUNAR/MARS	NEED	DATE	
	S.IW	ဟ	&E	٥	NSTS	NEED	DATE	
SIGNIFICANT MILESTONES:	1990 R&T BASE - COMPS, POWER SEMI'S	1991 [1992] ADV. DEV SSF, ALS	1995 DDT&E	∇ LEV. 5 MATURITY		VALIDATION NEAR COMPLETE:	 ADVANCED HIGH POWER PMAD CONCEPTS 	APPLICABLE TO CANDIDATE PROGRAMS

• ADVANCED MOTOR CONTROL ENABLING INDUCTION MOTOR EXPLOITATION

SPACE TRANSPORTATION AVIONICS TECHNOLOGY SYMPOSIUM FLIGHT ELEMENTS **IN - FLIGHT CREW TRAINING**



MAJOR MILESTONES (1990 - 1995)

- Test and evaluate faster machines with graphics capability
 - Test simulation interaction with current hardware
- · Develop non-realtime system with dynamics and collision detection on current hardware
 - Test and evaluate stereo graphics hardware
- · Investigate malfunction training concepts to establish viability

FACILITIES:

- Integraled Graphics Operations Analysis Laboratory (IGOAL)
 RMS MIL Simulators: Shuttle, SSF
- · Proximity Operations Simulators: Shuttle, Shuttle-C, OMV,
- JSC Systems Engineering Simulator

- C. Golt / JSC / FM8
- P. Galicki / JSC / FM8
- S. Murray / JSC / VG3

SPACE TRANSPORTATION AVIONICS TECHNOLOGY SYMPOSIUM **IN - FLIGHT CREW TRAINING** FLIGHT ELEMENTS

TECHNOLOGY ISSUES:

· Integration with existing flight systems

- Low weight, volume and power requirements Display and processor capabilities
- Provide for multiple trainees interacting within a realistic
 - synthetic 3-D training scenario
- Allow local storage of "digital" tapes of training scenarios
 - Facilities to upload from remote training library

CANDIDATE PROGRAMS:

- · Space Station Freedom (SSF)
- Remote Manipulator Systems (SRMS, SSRMS)
 - Flight Telerobotic Servicer (FTS)
- Orbital Maneuvering Vehicle (OMV) Piloting
 - Shuttle Piloting and Landing
 - Space Shuttle
- Remote Manipulator System
 - Proximity Operations
- Approach and Landing

SIGNIFICANT MILESTONES:

14 yrs

Development of System Tech and Demonstration of Prototypes Lunar Base Operational Implementation **Growth Station** Tech Base for Component Technologies

Exploration

- · Determine system requirements
- Develop training software for prototype system

MAJOR ACCOMPLISHMENTS:

- Development of kinematic and dynamic simulators for generic remote manipulator systems
- Teleoperated systems technology investigations - Helmet mounted display

 - Stereoscopic vision systems
- Manipulator Simulations: SRMS, SSRMS, FTS · Man-in-the-loop simulator development
 - · Spacecraft Simulators: Shuttle, OMV
- RMS Partial Task Trainer hosted on Silicon Graphics IRIS 4D/70GT
 - · Kinematic simulation with near real-time performance using low fidelity models
- RMS control panel and hand controllers

Lunar/Mars vehicle assembly node

Identify appropriate training tasks

- Define system architecture
- Develop and integrate system hardware
- · Flight demonstration of training capabilities
 - Operational trainer development
 - Training plan implementation

TECHNOLOGY HOLES

- o Modular S/W & H/W
- o Mathematical formulation
- o Long life reliable operation on-demand (inert or operational)
- o Validation methodology & tools (H/W & S/W)
- o Expert systems validation
- o Up-front requirements definition
- o GPS use abose 11 kn-m and during aerobraking
- o Antiquated technology replacement & down time (AAS)
- o Selection of communication approach for SSF prox. ops.
- o High data raate transmission to ground capability
- o Isolation/pointing/stabilization for optical communications
- o Total system integration & validation
- o Interconnect test beds across country
- o R&T and ADP funding inadequate to facilitate timely developments

CULTURAL CHANGES

- o All-weather launch capability and utilization
- o Launch with onboard defects/failures
- o Integration of flight systems and operations
- o Planning across multiple programs
- o Near-term user & technology insertion
- o Program selling minimizes use of technology
- o Trade information not data
- o Validation of only changes not total S/W package
- o Utilize commercial & other sectors for technology (eliminate NIH)
- o Paperless management
- o Reduction of standing ops armies

MAJOR DEMONSTRATIONS

o Fault tolerant avionics architecture for ALS - 5/90

o Fiber optics bus (lab demo) - 4 Gbits transmission

o Power system autonomy - 1990

o EMA (25-75 HP) demo for ALS - 1992

o Next-generation orbiter experimental cockpit facility needed